
Brayton Cycle(Joule's cycle):**The Ideal Cycle for Gas Turbine Engines**

- In reality, gas turbines operate on an open cycle
- Fresh air is continuously drawn into the compressor and exhaust gases are thrown out
- The open gas-turbine cycle can be modeled as a closed cycle
- The combustion process is replaced by a constant-pressure heat-addition process and the exhaust process is replaced by a constant-pressure heat-rejection process

Thermodynamic Analysis

- The four processes of the Brayton cycle are executed in steady-flow devices
- When changes in kinetic and potential energies are neglected, the energy balance for one of the processes can be expressed as

$$(q_{\text{in}} - q_{\text{out}}) + (w_{\text{in}} - w_{\text{out}}) = h_{\text{exit}} - h_{\text{inlet}}$$

Therefore, heat transfers to and from the working fluid are

$$q_{\text{in}} = h_3 - h_2 = c_p (T_3 - T_2)$$

$$q_{\text{out}} = h_4 - h_1 = c_p (T_4 - T_1)$$

Thermal Efficiency

The thermal efficiency of the ideal Brayton cycle under the cold-air-standard assumptions becomes

$$\begin{aligned}\eta_{\text{th, Brayton}} &= \frac{w_{\text{net}}}{q_{\text{in}}} = 1 - \frac{q_{\text{out}}}{q_{\text{in}}} \\ &= 1 - \frac{c_p (T_4 - T_1)}{c_p (T_3 - T_2)} \\ &= 1 - \frac{T_1 (T_4/T_1 - 1)}{T_2 (T_3/T_2 - 1)}\end{aligned}$$

EX: A gas turbine cycle has the following working parameters temperature of air entering the compressor at (27 °C) and (1.1 bar), the compressor ratio =4 and maximum temperature of cycle (1000 °C), find the thermal efficiency of the cycle?

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